

Physics-Based Radiator Design, Sizing & Weight Estimation Tool for Conceptual Design of More-, Hybrid-, and All-Electric Next Gen

Aircraft, Phase I

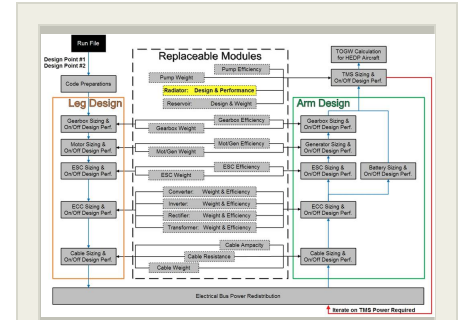
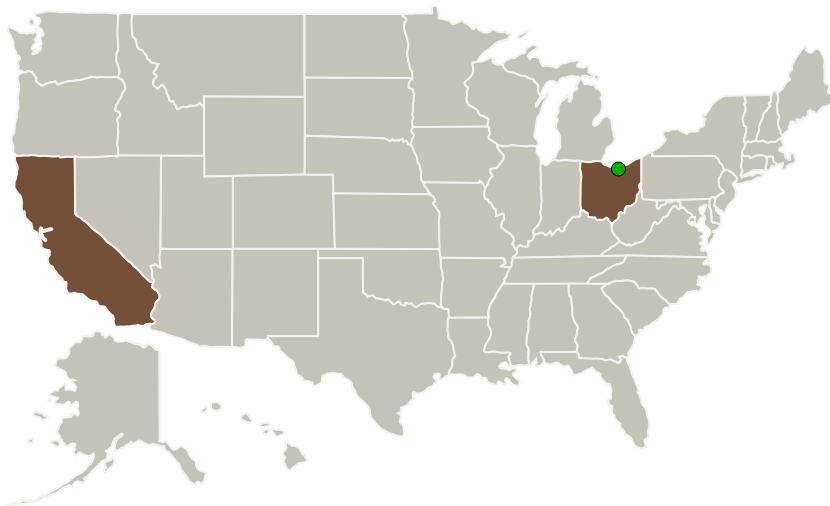
Completed Technology Project (2014 - 2014)



Project Introduction

Hybrid electric distributed propulsion (HEDP) systems have proven worthy for further consideration by approaching NASA's goals for N+2 and N+3 energy consumption, noise, emission and field length. The thermal management associated with these systems has been recognized as a major challenge to be overcome. ESAero's recent 2012 Phase I SBIR (NNX13CC24P) identified the radiator as a driving component within the thermal management system (TMS). Its design has profound first order effects on the weight, performance, and aerodynamic drag of the TMS, and second order effects on the weight and performance of the overall propulsion system. During the proposed Phase I SBIR, ESAero will upgrade the existing physics-based radiator design, analysis, and weight estimation conceptual design tool by improving the flexibility and fidelity of thermodynamic analysis and predicting the effects of integrating the radiator core within a well-designed duct. ESAero will call upon existing techniques to design a robust tool that more accurately predicts the "as-built" behavior of the component. These modifications are expected to dramatically improve the predicted weight and performance of the radiator and negate nearly all of the radiator drag by employing the Meredith Effect, as seen on the P-51 Mustang.

Primary U.S. Work Locations and Key Partners



Physics-based Radiator Design, Sizing & Weight Estimation Tool for Conceptual Design of More-, Hybrid-, and All-Electric Next Gen Aircraft Project Image

Table of Contents

| | |
|--|---|
| Project Introduction | 1 |
| Primary U.S. Work Locations and Key Partners | 1 |
| Project Transitions | 2 |
| Images | 2 |
| Organizational Responsibility | 2 |
| Project Management | 2 |
| Technology Maturity (TRL) | 3 |
| Technology Areas | 3 |
| Target Destinations | 3 |

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| Organizations Performing Work | Role | Type | Location |
|---|-------------------------|-------------|-------------------------|
| Empirical Systems Aerospace, Inc.(ESAero) | Lead Organization | Industry | Pismo Beach, California |
| ● Glenn Research Center(GRC) | Supporting Organization | NASA Center | Cleveland, Ohio |

Primary U.S. Work Locations

| | |
|------------|------|
| California | Ohio |
|------------|------|

Project Transitions

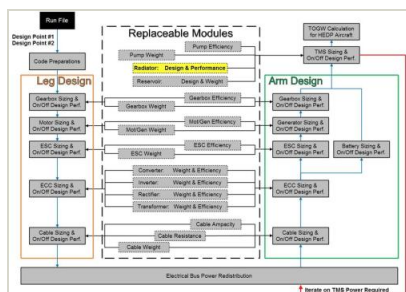
June 2014: Project Start

December 2014: Closed out

Closeout Documentation:

- Final Summary Chart(<https://techport.nasa.gov/file/137696>)

Images



Project Image

Physics-based Radiator Design, Sizing & Weight Estimation Tool for Conceptual Design of More-, Hybrid-, and All-Electric Next Gen Aircraft Project Image
(<https://techport.nasa.gov/image/133753>)

Organizational Responsibility

Responsible Mission Directorate:

Space Technology Mission Directorate (STMD)

Lead Organization:

Empirical Systems Aerospace, Inc. (ESAero)

Responsible Program:

Small Business Innovation Research/Small Business Tech Transfer

Project Management

Program Director:

Jason L Kessler

Program Manager:

Carlos Torrez

Principal Investigator:

Benjamin T Schiltgen

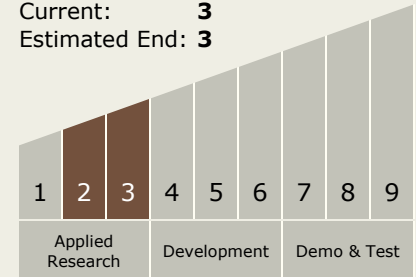
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Technology Maturity (TRL)

Start: **2**
Current: **3**
Estimated End: **3**



Technology Areas

Primary:

- TX14 Thermal Management Systems
 - └ TX14.2 Thermal Control Components and Systems
 - └ TX14.2.3 Heat Rejection and Storage

Target Destinations

The Sun, Earth, The Moon, Mars, Others Inside the Solar System, Outside the Solar System